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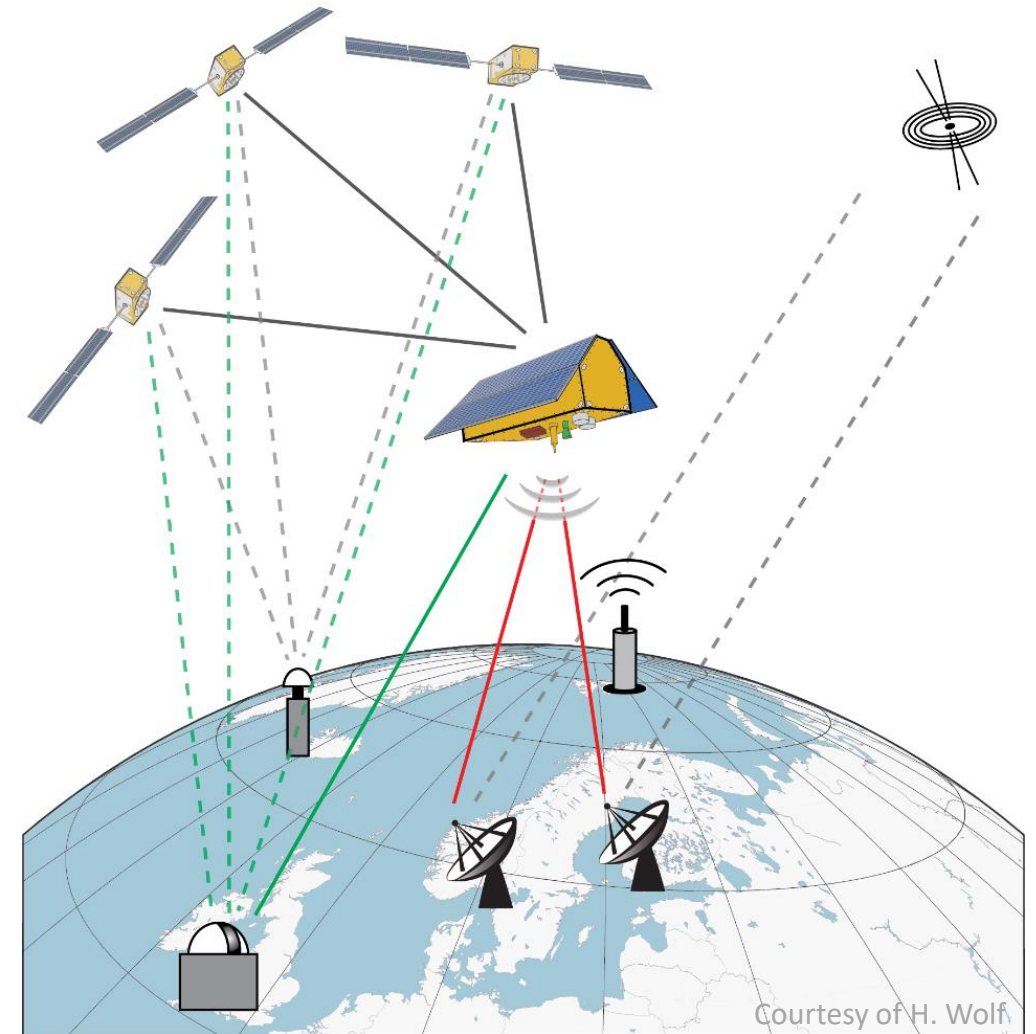
# Evaluating VLBI Scenarios for Genesis: Orbital and Observational Configurations

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# Genesis

- ESA mission to be launched in 2028
  - co-location satellite
  - circular orbit with
    - altitude: 6000 km
    - inclination:  $95^\circ$
- enhancement of ITRF through highly accurate space ties



# Motivation

**#1:** ongoing discussions regarding inclination

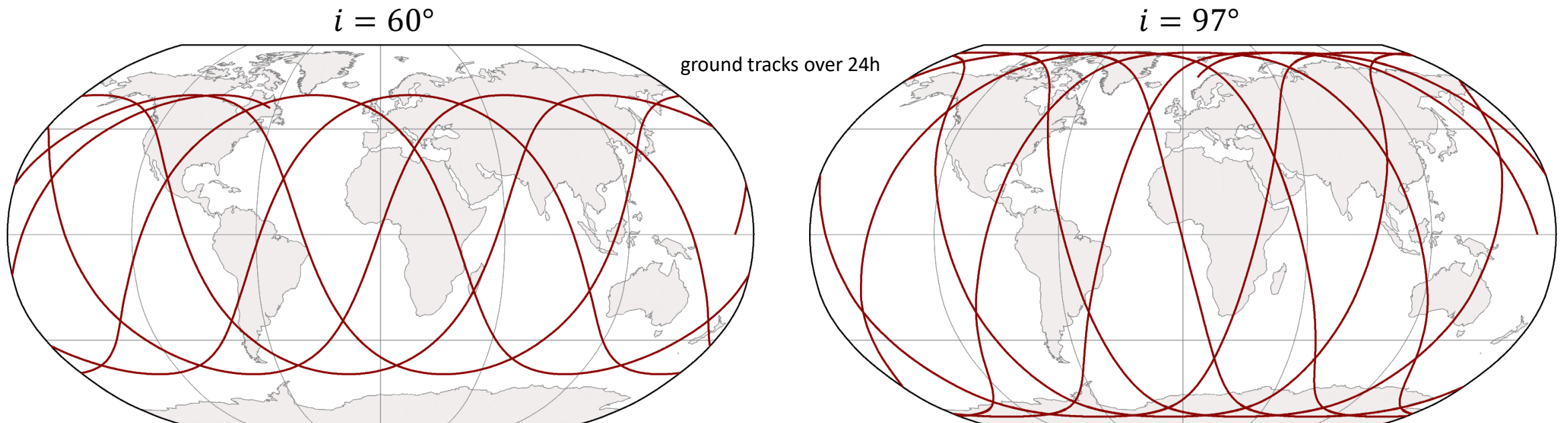
- inclined or polar orbit

- observing geometry and coverage

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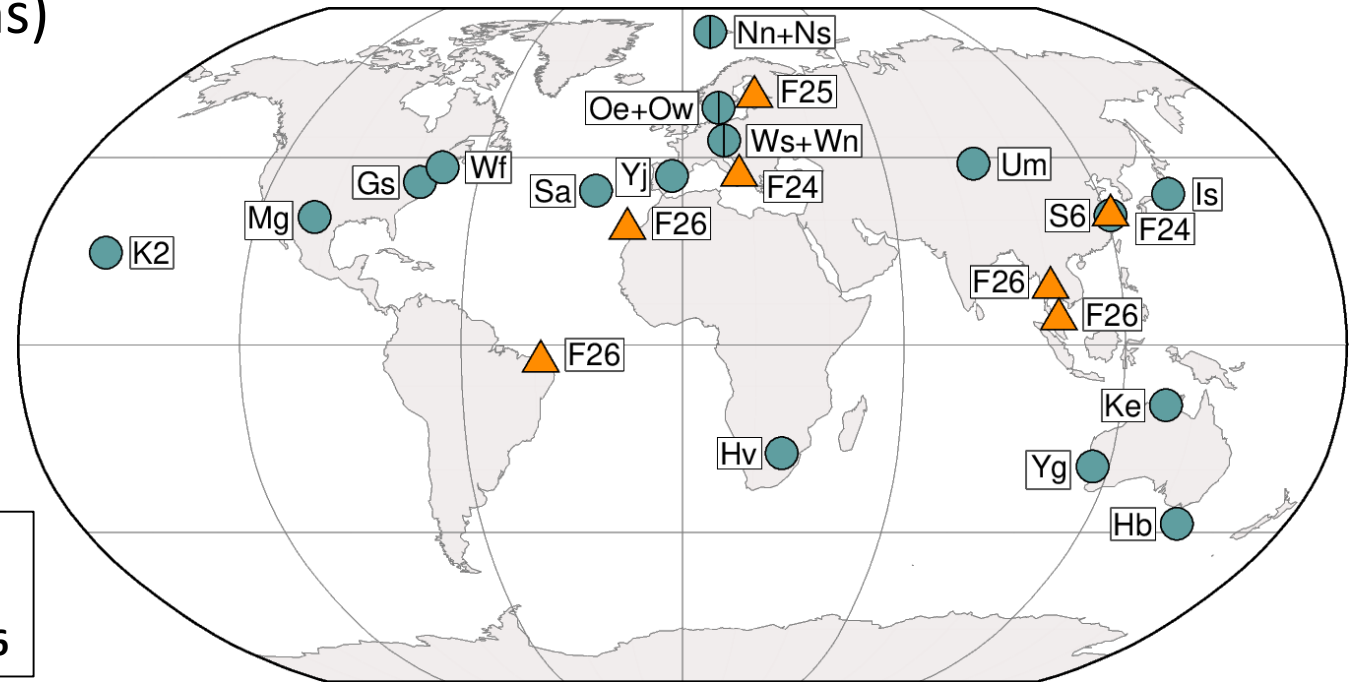
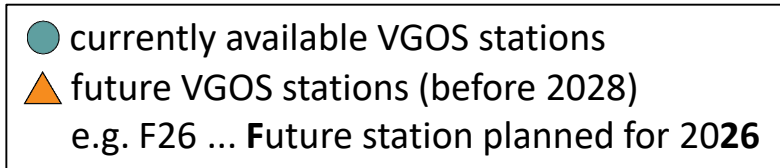


# Motivation

## #2: different network constellations

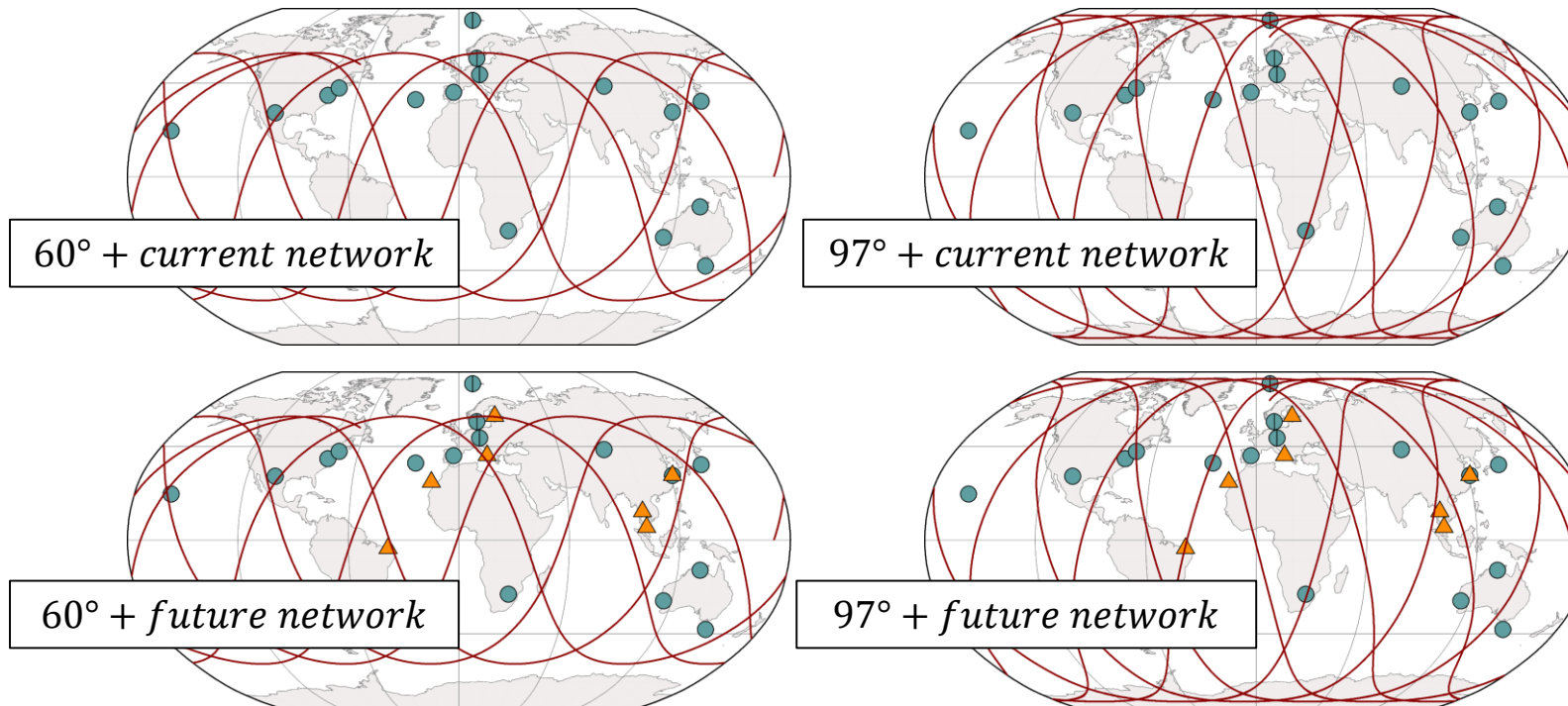
- current VGOS network (19 stations)
- future VGOS network (26 stations)

↓  
before 2028  
(launch of Genesis)



# Motivation

- examine the impact of different inclinations and station network configurations on the VLBI TRF → **4 scenarios**



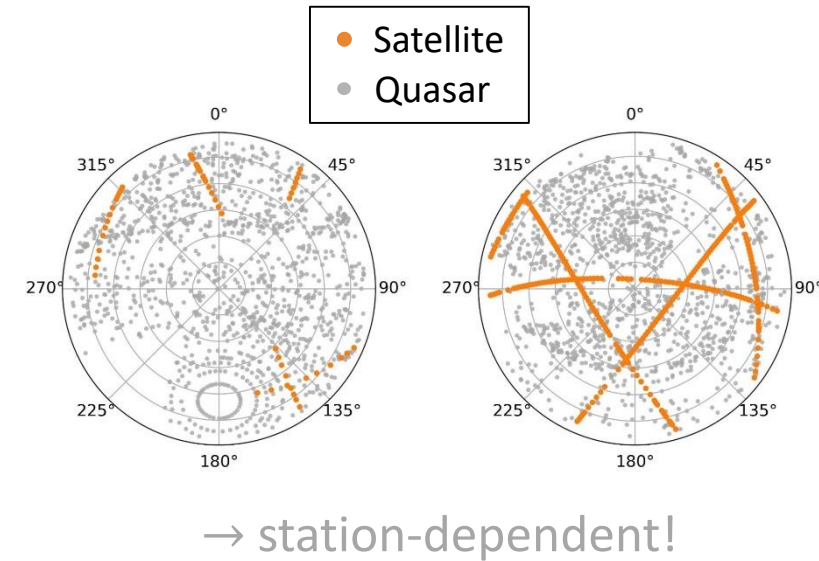
# Workflow (per scenario)

- scheduling
  - weekly 24-hour sessions over 2 years → 104 sessions
  - Genesis satellite orbit based on TLE file
  - scan length fixed to 30 seconds for Quasars and Satellite
  - constant weight for satellite
    - 15-20% of satellite scans w.r.t. all scans (on average)
- simulation
  - three main error sources (troposphere, clock, white noise)

|             |                                       |
|-------------|---------------------------------------|
| $C_n$       | $1.8 \times 10^{-7} \text{ m}^{-1/3}$ |
| white noise | 10 ps                                 |

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- single-session analysis
  - *estimation of:*
    - **station coordinates** [**S**]
    - tropospheric parameters [**Q** + **S**]
    - EOP [**Q**]
  - 104 solutions
- multi-session analysis
  - stacking of 104 NEQ to generate global position estimates (no velocities)

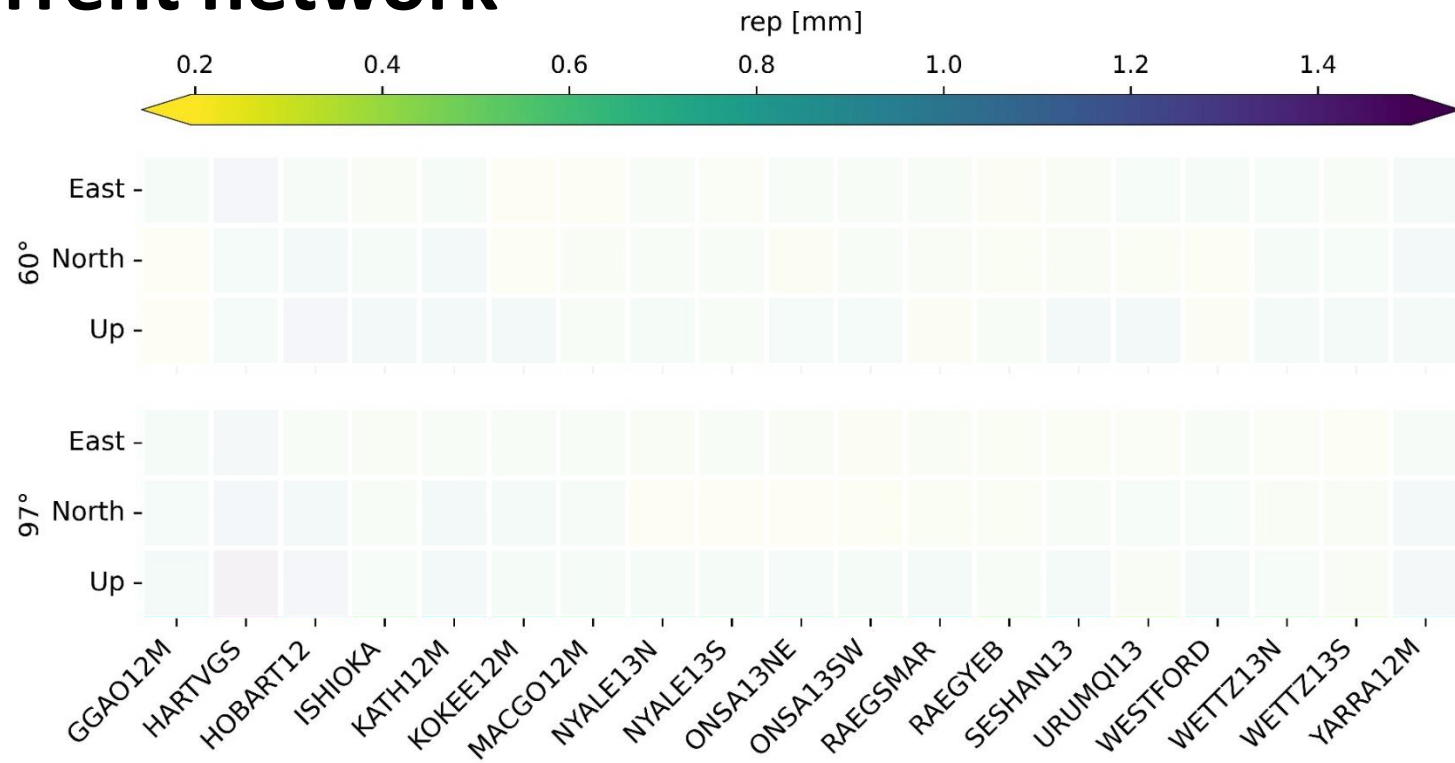
– *fixed to a priori values:*

- quasar coordinates
- satellite orbit

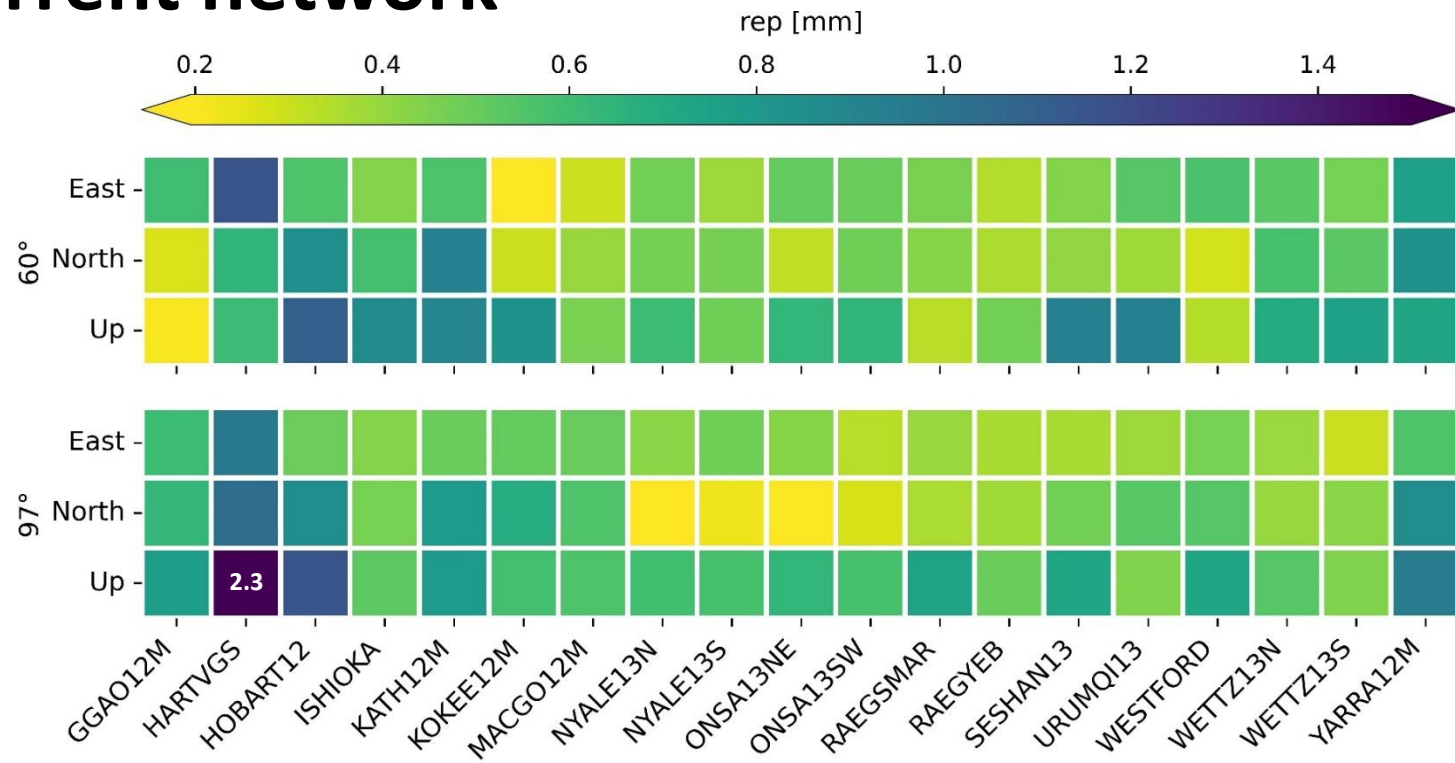
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    - 104 solutions
  - multi-session analysis
    - stacking of 104 NEQ to generate global position estimates (no velocities)
    - **10 global solutions**
-  *10 repetitions*
- *fixed to a priori values:*
    - quasar coordinates
    - satellite orbit
- **1 repeatability in ENU per scenario**

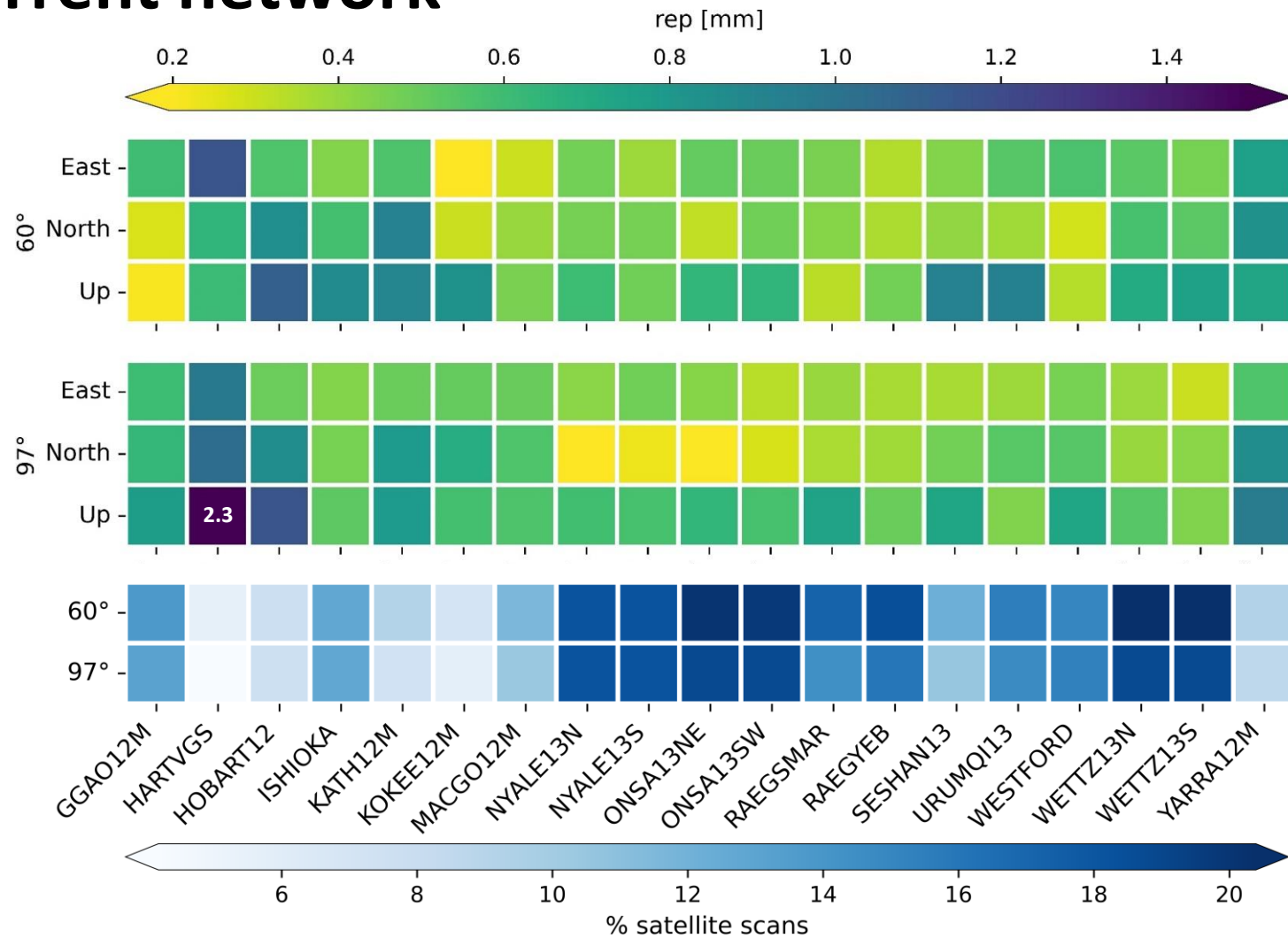
# Results – current network



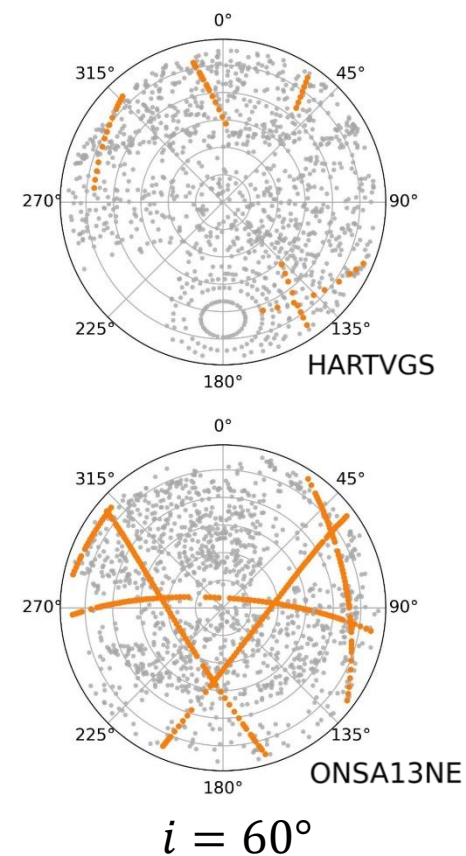
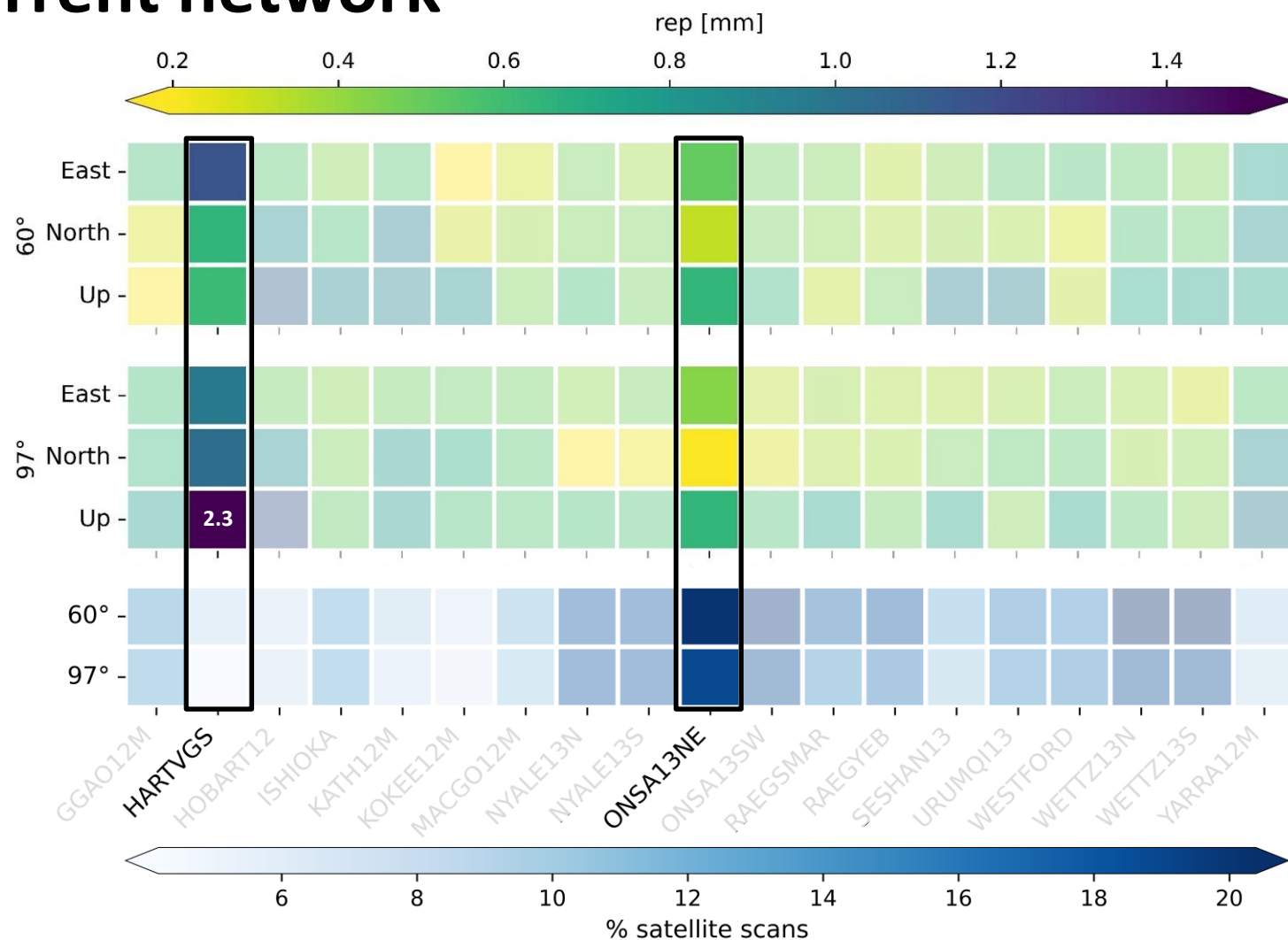
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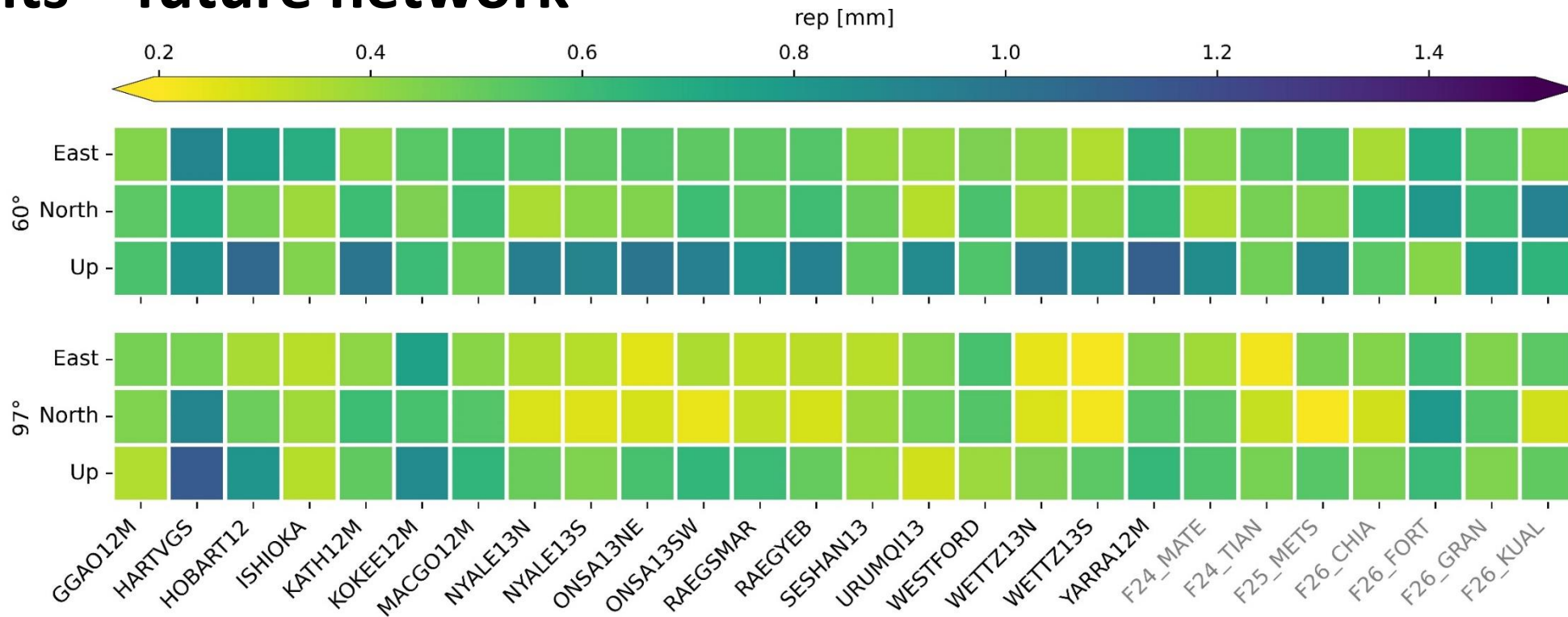
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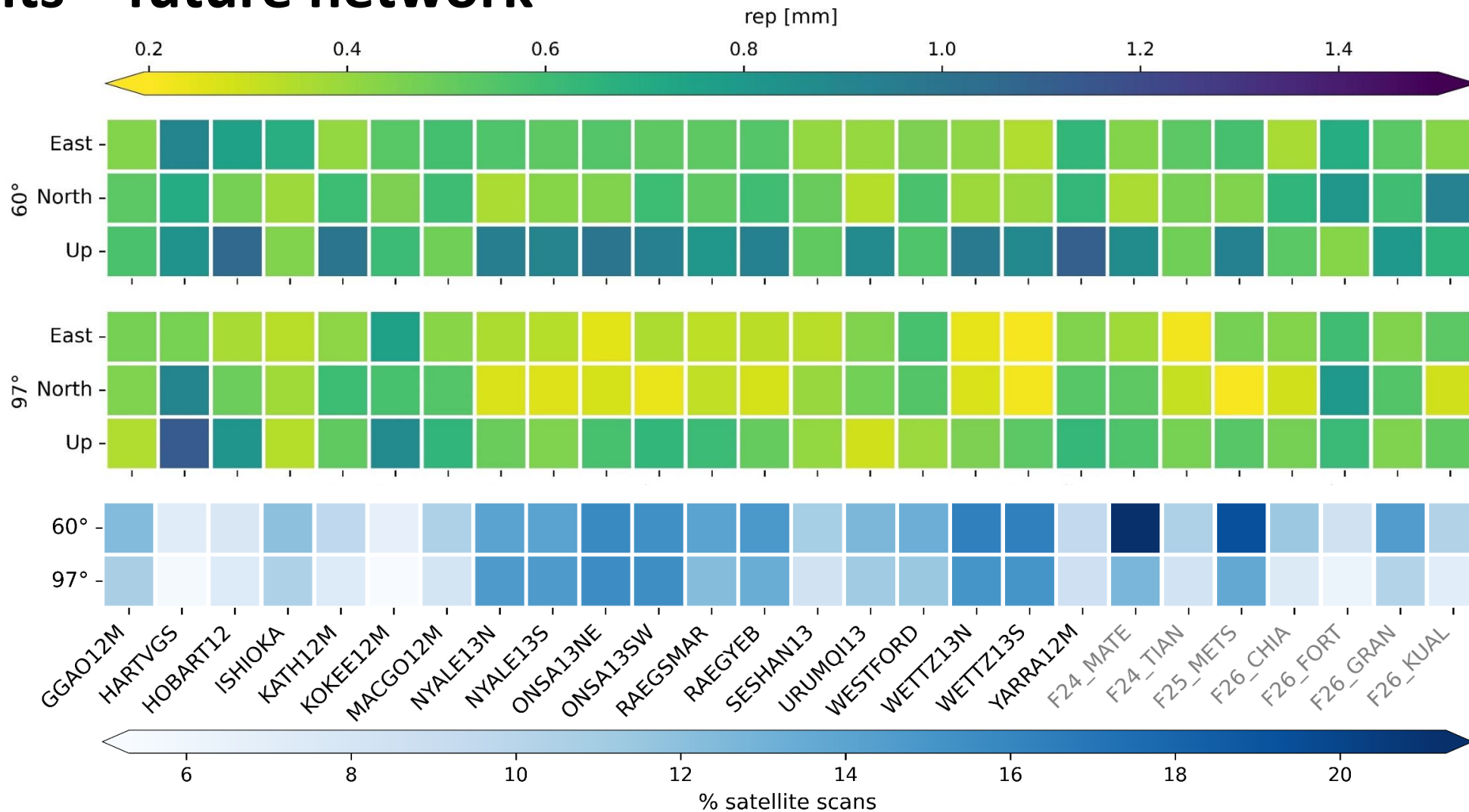
# Results – current network



# Results – future network

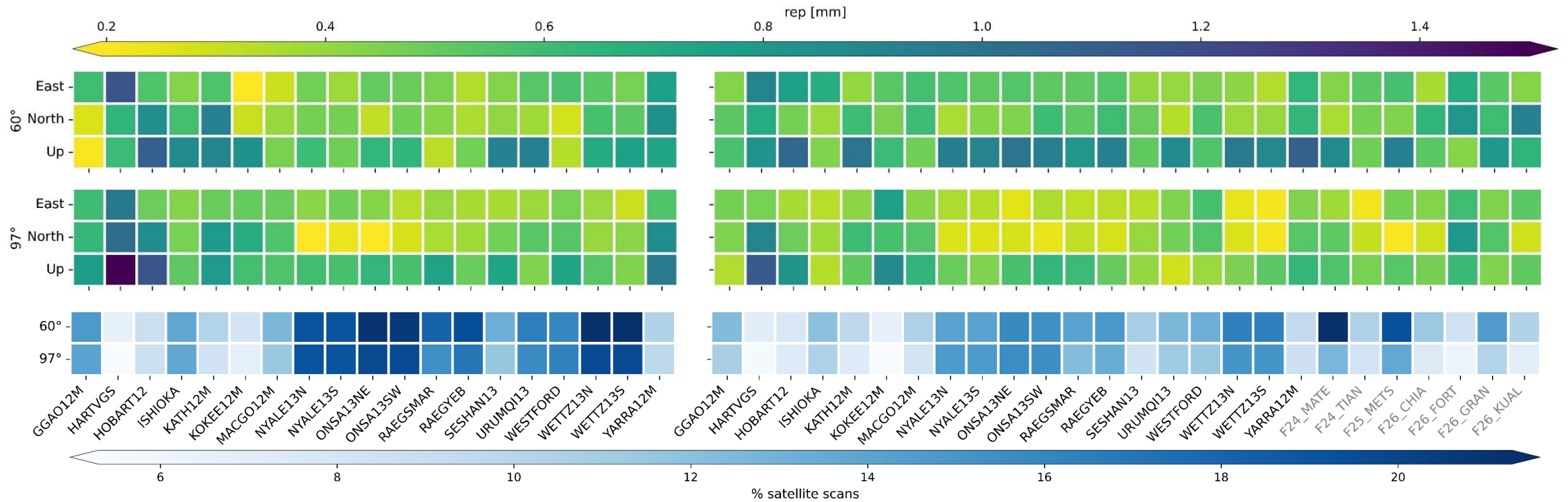


# Results – future network





# Results



# Conclusion

*limitations of this study → software-related and high computational cost*

- small improvement with polar orbit compared to inclined orbit
- additional improvement can be achieved through optimized scheduling
  - station-dependent weighting of satellite scans
  - future network achieves more scans overall, however no proportional increase in satellite scans is achieved → more satellite scans possible without degrading sky coverage

# References

- Wolf, H., Kern, L., Steinmetz, S., Böhm, J. (2025) Impact of the Inclination of Genesis on the VLBI Terrestrial Reference Frame. 27th Working Meeting of the European VLBI Group for Geodesy and Astrometry, EVGA2025, Matera, Italy. <https://doi.org/10.5281/zenodo.15209253>